SECTION 11—MATERIALS TECHNOLOGY

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OVERVIEW

Many classes of materials inherently have both military and commercial applications. While commercial applications are generally at lower performance levels than those of the military, this is not always the case. This section identifies those materials that provide specific military advantage and covers the physical properties, mechanical properties, behavior, and/or processing required to achieve that advantage. The technologies include materials engineered to defeat an enemy threat and functional materials needed to preserve the capability of highperformance hardware in daily operations. This section addresses seven categories of materials technology: The six listed above plus Biomaterials. The Biomaterials category addresses materials that function in biological application (e.g., medical implants) or are derived through biological types of processes (e.g., spider silk). This category is the subject of much R&D interest, but no militarily critical applications have been identified to date. Consequently, this category will not be discussed further in this volume. Armor materials include those materials and material systems specifically designed to protect equipment and personnel from enemy threats. Anti-Armor materials include those materials for projectiles used to defeat enemy armor. Materials for various types of penetrators, sabots, shaped charge liners, and their launchers are included. Because of the close interaction between the design of armor systems to protect against a threat and the design of systems to defeat armor, both the subcategories of materials technology are addressed under the combined Armor and Anti-Armor materials technology area. Superconducting electrical materials that provide the capability for lightweight, compact, highpower motors, magnets, and energy storage systems are addressed under Electrical Materials. Similarly, Magnetic Materials addresses a few specific materials with military applications in magnetic shielding, sonar, and high-speed power supplies. Materials critical to the reliable transmission of electromagnetic radiation to surveillance sensors, weapon guidance systems, or for countermeasures purposes while protecting the associated electronic componentry from the environment are addressed under the Optical Materials category. Structural Materials addresses a broad range of materials classes used for the fabrication of military systems. This category is subdivided into (1) highstrength materials, which encompass those materials used for fabrication of military vehicles of virtually every shape and description and (2) high-temperature materials, which are used primarily for propulsion purposes and hypersonic airframes. Those materials such as high-temperature lubricants, hydraulic fluids, anti-fouling coatings required for U.S. military hardware to operate reliably at superior levels of performance are addressed under the heading of Special Function Materials. Materials, systems, and arrangements used for Signature Control are addressed in Section 16.

SECTION 11.1—ARMOR AND ANTI-ARMOR MATERIALS

OVERVIEW

This subsection covers armor materials which includes metals and related composites, ceramics and related composites, and organic fibers and composites. Of special interest are ceramics that are near theoretical (> 98 percent) density, e.g., titanium diboride, boron carbide, and aluminum oxide; composite material; arrays of metal plates; ceramics; arrays of woven cloth; ceramics and metals; ceramics or single crystal whiskers in a bonded matrix; layers of metals; and high-explosive and very dense materials. Applications include body armor and vehicle armor protection of platforms/ vehicles, satellites, and tactical shelters. Anti-armor materials include steel, titanium, ceramics, and applique arrangements. Anti-armor materials of concern include forged or explosively formed or rolled molybdenum, tantalum, tungsten, and depleted uranium (DU).

Table 11.1-1. Armor and Anti-Armor Materials Militarily Critical Technology Parameters

| TECHNOLOGY | Militarily Critical Parameters Minimum Level to Assure US Superiority | Critical Materials | Unique Test, Production, and Inspection Equipment | Unique Software and Parameters | Control Regimes |
|---|--|-----------------------|---|---|--------------------------------|
| TUNGSTEN | Elongation >3% yield strength > 1250 MPa Ultimate tensile strength greater than 1270 MPa (185,900 psi) Density > 17.5 g/cm³. | Not applicable | Manufacturing, inspection and test equipment for projectile components and subcomponents | None identified | WA ML 18, 22 WA IL Cat 1 |
| COPPER, MOLYBDENUM OR TANTULUM SHAPED CHARGE LINERS | Density greater than 95% of theoretical | Not applicable | Shear/spin forming equipment capable of forming these materials to required tolerances | Programs to control forging and rolling of these materials | WA ML 18, 22 |
| DEPLETED URANIUM | In bar stock and as fabricated penetrators density greater than 18.0 gm/cm³, Yield strength greater than 850 MPa, Ultimate tensile strength greater than 1200 MPa; elongation > 20% | Not applicable | None identified | None identified | WA ML 18, 22 WA IL Cat 4 |
| SILICON CARBIDE | Density equal or greater than 98% theoretical | Not applicable | None identified | None identified | None |
| TITANIUM DIBORIDE | Density equal or greater than 98% theoretical | Not applicable | None identified | None identified | WA IL Cat 1 |
| BORON CARBIDE | Density equal or greater than 98% theoretical | Not applicable | None identified | None identified | WA ML 13, 12 |
| COMPOSITE MATERIALS SPECIALLY DESIGNED FOR KE ABSORPTION TO RESIST FRAGMENTATION OR TO IMPEDE SHOCK WAVE TRANSMISSION | Arrays of metal plate & low density foams or arrays of metal plates, ceramics and adhesive, or arrays of woven cloth, ceramics and metals; or Ceramics or single crystal whiskers in a bonded matrix; or Layers of metals and high explosives and very dense material, all in configurations specially designed to absorb kinetic energy, to resist fragmentation, to impede transmission of shock waves or to change the orientation of a projectile prior to penetration. Thus, each configuration is application-design specific. | Not applicable | None identified | None identified | WA ML 12, 13, 21 |

SECTION 11.2—ELECTRICAL MATERIALS

OVERVIEW

This subsection covers a limited group of superconducting materials in the form of wires, which permit very high field magnets and solenoids to perform at levels unattainable by other means.

Table 11.2-1. Electrical Materials Militarily Critical Technology Parameters

| TECHNOLOGY | Militarily Critical Parameters Minimum Level to Assure US Superiority | Critical Materials | Unique Test, Production, and Inspection Equipment | Unique Software and Parameters | Control Regimes |
|--------------------|---|-----------------------|--|--------------------------------|--------------------|
| MULTIFILAMENTARY | With lengths exceeding 100 m or | Nb | None identified | None identified | WA IL Cat 1 |
| SUPERCONDUCTIVE | with mass exceeding 100 g; and | | | | |
| COMPOSITE | (1) Embedded in a matrix other than | | | | |
| CONDUCTORS | a copper or copper-based mixed | | | | |
| CONTAINING | matrix; or | | | | |
| NIOBIUM-TITANIUM | (2) With a cross-sectional area less | | | | |
| | than 28 square micrometers. | | | | |
| SUPERCONDUCTIVE | With lengths exceeding 100 m or | Nb | None identified | None identified | WA IL Cat 1 |
| COMPOSITE | with mass exceeding 100 g; and | | | | |
| CONDUCTORS | (1) 9.85 K< T_c < 24 K at B = 0; | | | | |
| CONSISTING OF ONE | (2) With cross-sectional area | | | | |
| OR MORE | < 28 square micrometers; and | | | | |
| SUPERCONDUCTIVE | (3) Which remain in | | | | |
| FILAMENTS | superconducting state at $T = 4.2 \text{ K}$ | | | | |
| EXCLUDING NIOBIUM- | when exposed to a magnetic field | | | | |
| TITANIUM AND | corresponding to B = 12 T or | | | | |
| CERAMIC OXIDES | greater | | | | |
| SUPERCONDUCTIVE | With lengths exceeding 100 m and | None | None identified | None identified | None |
| COMPOSITE | $T_c > 24 \text{ K at B} = 0$; and | identified | | | identified |
| CONDUCTORS | which has "overall current density" | | | | |
| CONSISTING OF ONE | of 10,000 amps/square centimeter | | | | |
| OR MORE | exposed to a magnetic field | | | | |
| SUPERCONDUCTIVE | corresponding to $B = 2 T$ or greater | | | | |
| FILAMENTS | | | | | |
| CONTAINING | | | | | |
| CERAMIC OXIDES | | | | | |

SECTION 11.3—MAGNETIC MATERIALS

OVERVIEW

This subsection covers types and applications of magnetic materials that are militarily significant. These include (1) high relative permeability nickel-rich iron-nickel alloy sheets with 4 percent to 6 percent molybdenum; (2) magnetostrictive alloys, primarily rare earth iron alloys; and (3) both amorphous alloy strips and nanocrystalline alloy strips consisting of iron, cobalt, and/or nickel with boron, silicon, or phosphorous. Magnetic materials is an area where significant R&D activity is taking place in both nongovernment and government laboratories as well as abroad.

Table 11.3-1. Magnetic Materials Militarily Critical Technology Parameters

| TECHNOLOGY | Militarily Critical Parameters Minimum Level to Assure US Superiority | Critical Materials | Unique Test, Production, and Inspection Equipment | Unique Software and Parameters | Control Regimes |
|-----------------------------|--|--------------------|--|--------------------------------|--------------------|
| HIGH PERMEABILITY ALLOYS | Sheets with initial relative permeability (for fully annealed materials) of 120,000 or more;. Thickness of 0.05 mm or less; typically, but not limited to, Ni rich Fe-Ni with Mo concentrations of 4 to 6% | None identified | Hydrogen Anneal Equipment | None identified | WA IL Cat 1 |
| MAGNETOSTRICTIVE ALLOYS | With saturation magnetostriction > 500 ppm; or magnetomechanical coupling factor (k) > 0.8; primarily but not limited to rare earth alloys | None identified | None identified | None identified | WA IL Cat 1 |
| AMORPHOUS ALLOYS | Strips with a composition having at least 75 weight percent Fe, Co, or Ni; and a saturation magnetic induction (Bs) of 16 kG or more; Strip thickness of 0.02 mm or less; or an electrical resistivity of 0.0002 ohm-cm or more | None identified | None identified | None identified | WAIL Cat 1 |
| Nanocrystalline Alloys | Strips with a composition having at least 75 weight percent Fe, Co or Ni; and a grain size < 50 nm saturation magnetic induction (Bs) of 16 kG or more; and, a strip thickness of 0.02 mm or less; or an electrical resistivity of 0.0002 ohm-cm or more | None identified | None identified | None identified | WA IL Cat 1 |

SECTION 11.4—OPTICAL MATERIALS

OVERVIEW

This subsection includes mature material technologies for linear and nonlinear optical transmission in the visible and/or IR spectral regimes, including bulk materials and thin films and coatings. Special emphasis is placed on materials and coatings that are affordable, maintainable, and durable in the harsh environments experienced in military operations, such as the high-speed exposure to rain and dust and/or high temperatures and the high structural loads associated with high-speed, maneuvering flight. This subsection includes (1) IR Optical Materials, (2) IR coating materials for protection against hazardous environments, (3) germanium optics, (4) specialty transparent materials for coatings and filters, (5) nonlinear optical (NLO) materials for wavelength conversion, and (6) substrates and optical film for high energy laser optical components (mirrors, beamsplitters, windows).

Table 11.4-1. Optical Materials Militarily Critical Technology Parameters

| TECHNOLOGY | Militarily Critical Parameters Minimum Level to Assure US Superiority | Critical Materials | Unique Test, Production, and Inspection Equipment | Unique Software and Parameters | Control Regimes |
|--|---|--|---|---|-----------------------|
| IR OPTICAL ELEMENTS (I.E. BULK OPTICAL MATERIALS IN COMBINATION WITH APPROPRIATE COATINGS) | Transparent over any portion of the 1–12 microns spectral band and strength > 48 MPa (7 ksi), plus any of the following: plate size > 8 cm diameter, or domes of any size, or withstand 2 mm dia. raindrop impacts at 2.5 cm/hr rate at > Mach 1.0, or survive aerothermal heating rate > 100 w/cm² | ZnS and ZnSe > 100 cm³ in volume or > 8 cm diameter with thickness > 2 cm. | Equipment for rapid polishing of IR window materials. Single point diamond turning machines | None identified | WA IL Cat 6 MTCR 9 |
| IR COATING MATERIALS FOR PROTECTION AGAINST HAZARDOUS ENVIRONMENTS. | Rain and dust erosion; oxidation resistant at > 700 C; anti-reflection | Diamond-like carbon (DLC), diamond, MgO, ThF ₄ , BP, Ge | Equipment for measuring absolute reflectance to an accuracy of ± 0.1%. | None identified | WA Cat 6 |
| GERMANIUM (Ge) OPTICS | Resistivity < 15 ohm cm Absorption ≤ .03 cm ⁻¹ @ 10.6 microns; Diameter > 15 cm; Withstand blowing sand at 8 m/s wind velocity with particle density > 1.0 gm/m³ and particle size distribution 74–350 microns | Ge | Equipment for rapid polishing of IR windows and domes | None identified | |
| SPECIALTY TRANSPARENT MATERIALS FOR COATINGS/FILTERS | Selectable or variable bandpass or narrow rejection in 0.2–20 micron spectral region | Selected oxides and dielectrics (application dependent) | For control of coating deposition, especially thickness and composition. | Controls for deposition of coatings/ insitu characteriza- tion of coatings. | WA IL Cat 6 |

Table 11.4-1. Optical Materials Militarily Critical Technology Parameters (Continued)

| TECHNOLOGY | Militarily Critical Parameters Minimum Level to Assure US Superiority | Critical Materials | Unique Test, Production, and Inspection Equipment | Unique Software and Parameters | Control Regimes |
|--|---|---|---|---|--------------------|
| NONLINEAR OPTICAL (NLO) MATERIALS FOR WAVELENGTH CONVERSION | Spectral bandpass at 2–12 microns; average output power > 2 watts | ZnGeP ₂ ; KTA; GaSe; RTA CdGeAs ₂ ; RTA | None identified | None identified | WA IL Cat 6 |
| HEL OPTICAL COMPONENTS (MIRRORS, BEAMSPLITTERS, WINDOWS) | Substrates Diameter > 0.25 m for single crystal Si and SiC Low water fused silica < 100 ppm water Optical coating with total loss 200 ppm | Si, SiC low water fused silica coating materials ThF4, ZnSe, SiOZ, TiO ₂ , ZrO ₂ , Nb ₂ O ₅ , Al ₂ O ₃ | Single point diamond turning coating vacuum chambers Characterization equipment to measure absorptance and laser calorimetry), BRDF scatter reflectance | None identified | WA ML 19 |

SECTION 11.5—STRUCTURAL MATERIALS (HIGH-STRENGTH AND HIGH-TEMPERATURE)

OVERVIEW

Structural materials technology includes development, synthesis, processing, and characterization of a wide class of monolithic alloy and composite materials, as well as specialized coatings. Because the structural materials category is so broad, it has been subdivided into (1) high-strength materials used for fabrication of platforms, vehicles, and weapons and (2) high-temperature materials used primarily for propulsion. High-strength materials are very numerous, even for military systems, and include both ferrous and nonferrous metal alloys and both metal and polymer matrix composites. Included are maraging ultrahigh strength and high-strength low alloy steels; high-strength magnesium, aluminum and titanium alloys; aluminum-lithium alloys; and magnesium-, aluminum- and polymer-matrix composites, including both matrix and reinforcement constituents. Materials suitable for use at high temperature include iron-, nickel-, and cobalt-based superalloys and advanced intermetallics; tungsten and molybdenum alloys; oxide, carbide, nitride, and boride ceramics (in monolithic and composite forms); carbon-carbon composites, "high-temperature" aluminum alloys; and high-temperature protective coatings. This technology also serves as the base for emerging system technology advances. In general, structural materials have numerous dual-use applications.

Table 11.5-1a. Structural Materials (High-Strength and High-Temperature) Militarily Critical Technology
Parameters

| TECHNOLOGY | Militarily Critical Parameters Minimum Level to Assure US | Critical Materials | Unique Test, Production, and Inspection | Unique Software and Parameters | Control Regimes |
|----------------------------|---|-------------------------------------|---|--------------------------------|--------------------|
| | Superiority | | Equipment | | g |
| THERMOPLASTIC | The capability of having polymer | Aromatic liquid | None identified | Not included | WA IL Cat 1 |
| POLYMERIC RESINS | components, seals, or composite | crystal polymers, | | | |
| FOR ELEVATED | structures or laminates that can | polyarylene | | | |
| TEMPERATURE | operate at temperatures ≥ 523 °K | ketones, | | | |
| APPLICATIONS | (250 °C) | polyarylene ether | | | |
| | | ketones, | | | |
| | | polyarylene | | | |
| | | sulfides, | | | |
| | | polyarylene | | | |
| | | sulfones, | | | |
| | | polyarylene ether | | | |
| | | sulfones and other | | | |
| | | polyarylene | | | |
| _ | | materials | | | |
| FLUORINATED AND | The ability to perform at | Fluorinated and | None identified | None identified | WA IL Cat 1 |
| NONFLUORINATED | temperatures ≥ 205 °C | nonfluorinated | | | |
| POLYMERIC | (400 °F) in operation | polymers; | | | |
| MATERIALS FOR | | bismaleimides; aromatic | | | |
| THE MATRICES OF | | | | | |
| COMPOSITE STRUCTURES OR | | polyamide-imides; aromatic | | | |
| LAMINATES | | polyimides; | | | |
| LAMINATES | | aromatic | | | |
| | | polyetherimides | | | |
| RESIN | Specific tensile strength ≥ 23.5 x | Any fiber above | Prepregging | Not included | WA IL Cat 1 |
| IMPREGNATED | 10 ⁴ m and | AS4/T-500 and | equipment, | | |
| FIBER PREPREGS | Specific modulus ≥13.87 x 10 ⁶ m | matrix $T_a > 145 ^{\circ}\text{C}$ | fibers, fiber | | |
| UTILIZING FIBROUS | for fibers or filaments | after cure | treatment, | | |
| OR FILAMENTARY | T _a > 145 °C after cure (as | | resins and | | |
| REINFORCEMENTS | determined by ASTM D4065 or | | combination of | | |
| | national equivalents) | | all of these. | | |

Table 11.5-1a. Structural Materials (High-Strength and High-Temperature) Militarily Critical Technology Parameters (Continued)

| TECHNOLOGY | Militarily Critical Parameters Minimum Level to Assure US Superiority | Critical Materials | Unique Test, Production, and Inspection Equipment | Unique Software and Parameters | Control Regimes |
|---|--|--|--|--|---------------------|
| FIBROUS OR FILAMENTARY POLYMERIC MATERIALS | Organic fibers or filaments with specific tensile strength > 3.6 GPa and tensile modulus > 130 GPa and for use above 330 C (540 F) | None identified | None identified | None identified | WA IL Cat 1 |
| COMPOSITE MATERIALS OR LAMINATES THAT CONTAIN CONTINUOUS CARBON OR INORGANIC FIBROUS OR FILAMENTARY MATERIALS WITH A POLYMER MATRIX | Fibers with specific modulus of 13.87×10^6 m and specific strength of > 23.05×10^4 m; matrix T_g > 145 °C after cure. | Any fiber with properties indicated under militarily critical parameters | All equipment associated with the production of high performance organic matrix composite components and structures | Not included | WA IL Cat 1 |
| ALUMINUM ALLOYS (PM) | Ultimate tensile strength of > 240 MPa (35 ksi) at 200 °C, 415 MPa (60 ksi) at 25 °C, with specified powder parameters; of composition Al-Mg-X; Al-X-Mg; Al-Zm-X; Al-X-Zm; Al-Fe-X; Al-X-Fe made by powder metallurgy. | None identified | Equipment to carry out process specified in WAWG 2 Item 1.C.2.b.2. Equipment for producing and processing contamination free alloy powder. | Not included | WA IL Cat 1 |
| ALUMINUM AND ALUMINUM ALLOY MATRIX COMPOSITES | Stiffness > 103 GPa (15 Msi) and Fracture toughness > 297 MPa root m (18 ksi) root in. | Reinforcement fibers and protective fiber coatings | None identified | Process control software | WA IL Cat 1 |
| TITANIUM (BURN- RESISTANT) SUCH AS PRATT & WHITNEY "ALLOY C" (PROPRIETARY COMPOSITION), OR EQUIVALENT. | A temperature that supports sustained combustion 220 °C (400 °F) higher than that of traditional structural titanium alloys | None identified | Procedures and equipment designed for contamination free processing of alloys and powders or for superplastic forming and diffusion bonding. | Not included | None |
| TITANIUM-MATRIX COMPOSITES | Tensile strength ≥ 1100 MPa (160,000 psi) Elastic modulus ≥ 172 GPa (25 Msi) and Density ≤ 4.4 x 10³ kg/m³ (0.16 lb/in. ³). | Reinforcement fibers and titanium alloy foil | Hot isostatic pressing and canning equipment. | Process control software, consolidation models, and fiber spacing procedures. | WA IL Cat 1 |
| HIGH THERMAL CONDUCTIVITY CARBON FIBERS | Thermal conductivity Ka ≥ 1000 watts/mC | None identified | None identified | None identified | None (Continued) |

Table 11.5-1a. Structural Materials (High-Strength and High-Temperature) Militarily Critical Technology Parameters (Continued)

| TECHNOLOGY | Militarily Critical Parameters Minimum Level to Assure US Superiority | Critical Materials | Unique Test, Production, and Inspection Equipment | Unique Software and Parameters | Control Regimes |
|---|---|---|---|--------------------------------|--------------------|
| TWO DIMENSIONALLY (2D) REINFORCED CARBON FIBER CARBON MATRIX COMPOSITES | Specific modulus exceeding 10.15 x 10 ⁶ m; and Specific tensile strength exceeding 17.7 x 10 ⁴ m | Fibers, including pitch, PAN, high thermal conductivity fibers, high modulus fibers | Densification, weaving, automated fabrication, process control, high temp. sensors, processing (CVD, CVI, pitch impregnation, copyrolysis). | Not included | WA IL Cat 1 |
| COMPOSITE MATERIALS OR LAMINATES THAT CONTAIN CONTINUOUS CARBON OR INORGANIC FIBROUS OR FILAMENTARY MATERIALS WITHIN METAL OR CARBON MATRICES | Carbon fibers with specific tensile strength > 17.7 x 10 ⁴ m and Specific modulus > 10.15 x 10 ⁵ m; Inorganic fibers with specific tensile modulus > 2.54 x 10 ⁶ m and Melting/ softening/ decomposition point > 1650 °C | Any fiber with properties indicated under militarily critical parameters | Fiber production, fiber treatments, carbon and metal infiltration equipment. | Not included | WA IL Cat 1 |
| ULTRA HIGH STRENGTH/HIGH TOUGHNESS STEEL | Yield strength > 1.965 GPa (285 ksi) toughness > 111 MPa root in. (100 ksi root in) | None identified | "AerMet 100," patented by Carpenter Technology Co. | None identified | None |
| INORGANIC FIBERS AND FILAMENTARY MATERIALS | Having specific modulus > 2.54 x 10 ⁶ m (1650 °C) and A melting or decomposition point > 1,922 K in an inert environment | None identified | Equipment for converting polymeric fibers into carbon or silicon carbide fibers; equip for chemical vapor deposition; equipment for wet spinning of refractory ceramics and equip for converting aluminum to alumina fibers | Not included | WA IL Cat 1 |

Table 11.5-1b. Structural Materials (High-Strength and High-Temperature) Militarily Critical Technology Parameters

| TECHNOLOGY | Militarily Critical Parameters Minimum Level to Assure US Superiority | Critical Materials | Unique Test, Production, and Inspection Equipment | Unique Software and Parameters | Control Regimes |
|---|---|--|---|---|--------------------|
| TITANIUM ALLOYS (MADE FROM METAL ALLOY POWDER OR PARTICULATES OR CAST OR WROUGHT) | Advanced propulsion titanium alloys; stress rupture life of 10,000 hours or longer at 450 °C at a stress of 200 MPa | Titanium alloy powder made in controlled environment from melt or mechanical alloying | Vacuum, gas or rotary atomization, splat quenching, melt spinning and comminution, mechanical alloying or plasma spray, cospray or osprey process | Not included | WA IL Cat 1 |
| TITANIUM ALUMINIDES | Either ductility of > 0.5% elongation at 25 °C or Tensile strength of > 250 MPa at 870 °C and Containing >10 weight % of aluminum; in crude or semi-fabricated forms, containing at least one additional alloying element | Not applicable | Vacuum, gas or rotary atomization, splat quenching, melt spinning and comminution, mechanical alloying or plasma spray, cospray or osprey process | Not included | WA IL Cat 1 |
| TITANIUM-ALUMINIDE MATRIX COMPOSITES | Tensile strength > 1172 MPa at 540 °C, Creep resistance > 300 h at 634 MPa at 540 °C and >10,000 cycle low cycle fatigue life at 650 °C and 634 MPa | Reinforcement fibers, aluminide foil and contamination- free aluminide powder | Hot isostatic pressing and canning equipment. | Process control software | WA IL Cat 1 |
| NICKEL-BASED ALLOYS MADE FROM METAL ALLOY POWDER OR PARTICULATE MATERIAL | Stress rupture life > 10,000 hours at 550 MPa (80 ksi), 650 °C, or A low cycle fatigue life of 10,000 cycles or more at 550 °C at a maximum stress of 700 MPa | Nickel alloy powder made in a controlled environment from the melt or by mechanical alloying | Vacuum, gas or rotary atomization, splat quenching, melt spinning and comminution, mechanical alloying or plasma spray, cospray or osprey process | Not included | WA IL Cat 1 |
| BULK GRAPHITES | Bulk density of at least 17.2 g/cc measured at 15 °C and having a particle size of 100 mm or less | Not applicable | None identified | None identified | MTCR 8 |
| DEVELOPMENT OR PRODUCTION OF PYROLYTICALLY DERIVED MATERIALS | Formed on mold or mandrel from precursor gases which decompose in 1,300 to 2,900 °C temp and range at pressure of 1 mm to 150 mm | Not applicable | Specifically designed nozzles for processing | Software for development, production and use; know-how | MTCR 7 |

Table 11.5-1b. Structural Materials (High-Strength and High-Temperature) Militarily Critical Technology Parameters (Continued)

| | Parameters (Continued) | | | | | |
|--|---|---|--|---|-----------------------|--|
| TECHNOLOGY | Militarily Critical Parameters Minimum Level to Assure US Superiority | Critical Materials | Unique Test, Production, and Inspection Equipment | Unique Software and Parameters | Control Regimes | |
| CERAMIC-CERAMIC COMPOSITE MATERIALS WITH MATRICES OF SILICON—, ZIRCONIUM—OR BORON CARBIDE OR NITRIDE, REINFORCED WITH FIBERS FROM THE C, SI-N, SI-C, SI-C-O- N, SI-AL-O-N, OR SI-O-N SYSTEMS OR MATRICES OF ALUMINUM OXIDE OR ALUMINUM NITRIDE, REINFORCED WITH ALUMINA FIBERS | Tensile strength > 105 MPa (15 ksi) at 1000 °C | Not applicable | NDE methods still in development | Not included | WA IL Cat 1 | |
| CERAMIC-CERAMIC COMPOSITE MATERIALS WITH A GLASS OXIDE- MATRIX REINFORCED WITH FIBERS FROM ANY OF THE FOLLOWING SYSTEMS: SI-N, SI- C, SI-AL-O-N; OR SI-O-N; C, OR "NICALON" | Tensile strength > 105 MPa (15 ksi) at 1000 °C | B-Mg-Al silicate (Compglas) with C or silicon carbide fibers; "Blackglass" (SiO ₂ + C) | None identified | None identified | WA IL Cat 1 | |
| 3 DIMENSIONALLY REINFORCED (3D) CARBON FIBER- CARBON MATRIX COMPOSITES AND HIGHER LEVELS OF REINFORCED ("N"D) COMPOSITES | Specific modulus exceeding 10.15 x 10 ⁶ m; and Specific tensile strength exceeding 17.7 x 10 ⁴ m | Carbon fibers and graphite fibers including pitch, PAN, and rayon. Pitch and resin matrix systems. | High pressure densification, weaving (less than 0.030"), automated fabrication, processing equipment, computed tomography, inspection capability; impregnation equipment | Process controls (i.e., weaving, densification, time, temperature, pressure) and know-how | WA IL Cat 1 MCTR 8 | |
| FIBROUS OR FILAMENTARY MATERIALS FOR USE ABOVE 540 C (1000 °F) | Specific modulus exceeding 12.7 x 10 ⁶ m; and A specific tensile strength exceeding 23.5 x 10 ⁴ m; Carbon fibrous or filamentary materials with a specific modulus exceeding 12.7 x 10 ⁶ m; and A specific tensile strength exceeding 23.5 x 10 ⁴ m | Not applicable | Ultrasonic inspection equipment | Not included | WA IL Cat 1 | |

Table 11.5-1b. Structural Materials (High-Strength and High-Temperature) Militarily Critical Technology Parameters (Continued)

| TECHNOLOGY | Militarily Critical Parameters Minimum Level to Assure US Superiority | Critical Materials | Unique Test, Production, and Inspection Equipment | Unique Software and Parameters | Control Regimes |
|---|--|--------------------|---|--------------------------------|-----------------------------------|
| METAL ALLOY AND INTERMETALLIC ALLOY POWDER PRODUCTION, HANDLING AND CONSOLIDATION | Powder metals with contamination levels < 3 non-metallic particles larger than 100 microns per billion metal particles | None identified | Equipment for producing and processing contamination-free alloy powdering. | Process control software | WA IL Cat 1 |
| SINGLE CRYSTAL METALS AND INTERMETALLICS | Incipient melting points above 1090 °C | Not applicable | Equipment for single crystal blade growth; NDE/NDI equipment | Not included | WA IL Cat 1 |
| THERMAL PROTECTION COATINGS | Temperature at metal-coating interface > 1150 °C (2100 °F) | None identified | Plasma spray, CVD, e-beam, deposition, laser ablation (see Section 10) | Not included | WA ML 10 WA IL Cat 1 MTCR 7 |

SECTION 11.6—SPECIAL FUNCTION MATERIALS

OVERVIEW

The special function materials technology area covers a broad range of materials (over 100) for equally numerous applications. Included are hydraulic fluids and seals; turbine engine lubrication fluids and seals; high-temperature solid lubricants; protective paints and coatings for corrosion resistance; antifouling coatings for ships; rain and sand erosion-resistant coatings for missile domes; electronic cooling fluids; fuel system seals and sealants; gyro floatation fluids; and fire retarding materials. This technology area is included in current DoD R&D programs and is an area of significant activity in government and nongovernment laboratories.

Table 11.6-1. Special Function Materials Militarily Critical Technology Parameters

| TECHNOLOGY | Militarily Critical Parameters Minimum Level to Assure US Superiority | Critical Materials | Unique Test, Production, and Inspection Equipment | Unique Software and Parameters | Control Regimes |
|--|--|----------------------------------|---|--------------------------------|--|
| CHEMICAL/ BIOLOGICAL AGENT RESISTANT COATINGS | Specific formulations for substrate material/agent pairs with at least 95% of the coating being saturated elastomeric polymers | None identified | None identified | None identified | WA IL Cat 1 |
| EROSION PROTECTIVE COATINGS | Resistance to rain and sand erosion coatings composed of elastomeric polymers | Polyurethanes and fluoropolymers | None identified | None identified | WA IL Cat 1 |
| FLUORINATED POLYMERS | Elastomeric polymers containing 10% or more of combined fluorine (by weight) | None identified | Very specialized polymerization facilities, reactor vessels, or reactors with or without agitators and suitable agitators with total internal volume > 100 liters and less than 20,000 liters | None identified | WA IL Cat 1 WA ML 8 MCTR 10, 11 |
| COMPLETELY FLUORINATED POLYMERS | Fluoroelastomers containing no hydrogen | None identified | Very specialized polymerization facilities and knowhow required to manufacture these materials | None identified | WA IL Cat 1 WA ML 10 MTCR 3, II |
| COPOLYMERS OR TERPOLYMERS MADE FROM VINYLIDENE FLUORIDE, HEXA- FLUOROPROPYLENE, TETRA- FLUOROETHYLENE OR PERFLUORO- METHYLVINYLETHER | Elastomeric polymers containing 10% or more of combined fluorine (by weight) | None identified | Very specialized polymerization facilities, reactor vessels, or reactors with or without agitators and suitable agitators with total internal volume > 100 liters and less than 20,000 liters | None identified | WA IL Cat 1 MTCR 10, 11, 17 |
| CHLORO- FLUOROCARBONS | No flash point; autogenous ignition temperature > 977 K (704 °C) or higher; contains only chlorine, hydrogen and carbon | None identified | None identified | None identified | WA IL Cat 1 MTCR 10, 11 |

Table 11.6-1. Special Function Materials Militarily Critical Technology Parameters (Continued)

| | | | | , | |
|--|--|--------------------|--|--------------------------------|-------------------------------|
| TECHNOLOGY | Militarily Critical Parameters Minimum Level to Assure US Superiority | Critical Materials | Unique Test, Production, and Inspection Equipment | Unique Software and Parameters | Control Regimes |
| FLUORINATED PHOSPHAZINE POLYMERS | Elastomers containing 30% or more by weight of combined fluorine | None identified | Very specialized polymerization facilities, reactor vessels, or reactors with or without agitators and suitable agitators, with total internal volume > 100 liters and < 20,000 liters | None identified | MTCR 10, 11 |
| SYNTHETIC HYDROCARBONS | Flashpoint > 477 K (204 °C); pour point 239 K (-34 °C) or lower; viscosity index 75 or greater; thermal stability to 616 K (343 °C) or greater | None identified | None identified | None identified | WA IL Cat 1 MTCR 10, 11 |
| SILAHYDROCARBONS | Contains only silicon, hydrogen and carbon, flash point > 477 K (204 °C); pour point 239 K (–34 °C) or lower; viscosity index 75 or greater, thermal stability to 616 K (343 °C) or greater | None identified | None identified | None identified | WA IL Cat 1 MCTR 10, 11 |
| PERFLUORO- POLYALKYL ETHER BASED LUBES AND GREASES | Viscosity of 1.0 centistoke or more at 343 °C (600 °F) and 20,000 centistokes or more at -51 °C; Oxidatively stable at 316 °C (per ASTM D4636 or national equivalent) | None identified | None identified | None identified | MTCR 10, 11 |
| POLYOL ESTER BASED LUBES | 4 centistokes minimum viscosity at 100 °C and 20,000 centistokes maximum at –51 °C; Oxidatively stable at 220 °C (per ASTM 4634 or national equivalent). | None identified | None identified | None identified | MCTR 10, 11 |
| PHENYLENE ETHER BASED LUBES AND GREASES [PURE OR MIXED WITH ALKYLPHENYLENE ETHER(S) OR THIOETHER(S)] | Provide lubrication over - 54 °C to + 2600 C (-65 °F to + 500 °F) | None identified | None identified | None identified | WA IL Cat 1 |
| FLUORINATED SILICONE FLUIDS | Kinematic viscosity less than 5000 centistokes at 25 °C | None identified | None identified | None identified | WA IL Cat 1 |
| THIO ETHERS [PURE OR MIXED WITH PHENYLENE AND/OR ALKYLPHENYLENE ETHER] | Provide lubrication over -54 °C to + 2600 C (-65 °F to + 500 °F) | None identified | None identified | None identified | WA IL Cat 1 |
| POLYTHIOETHER BASED SEALANTS | Seal over broad temperature range –54 °C to 150 °C (-65 °F to 300 °F) | None identified | None identified | None identified | None (Continued) |

Table 11.6-1. Special Function Materials Militarily Critical Technology Parameters (Continued)

| TECHNOLOGY | Militarily Critical Parameters Minimum Level to Assure US Superiority | Critical Materials | Unique Test, Production, and Inspection Equipment | Unique Software and Parameters | Control Regimes |
|--|--|--------------------|--|--------------------------------|------------------------------------|
| SOLID LUBRICANT MATERIALS | Coefficient of friction 0.12 maximum; meet the adhesion criteria of ASTM D2510; outgassing mass loss 1.0% maximum; volatile condensible material 0.1% maximum, as determined per ASTM E595; or coefficient of friction of 0.25 maximum from 4 °C to 538 °C (+ 40 °F to 1,000 °F) as per ASTM D2981; oxidatively stable at 538 °C (1000 °F) minimum; load capacity of 1.25 MPa (150,000 psi) maximum as per ASTM E595 | None identified | None identified | None identified | MTCR 9, 11 |
| FLUORINATED FLUIDS FOR COOLING | Completely (100%) fluorinated | None identified | None identified | None identified | WA ML 15 WA IL Cat 1 MTCR 11 |
| DIBROMOTETRA- FLUOROETHANE OR POLYCHLOROTRI- FLUOROETHYLENE OR POLYBROMOTRI- FLUOROETHYLENE BASED, FULLY BROMINATED OR FULLY CHLORINATED HIGH DENSITY FLUIDS | Contains at least 85% of the named compounds and having a purity exceeding 99.8% and Containing less than 25 particles (200 micrometers or larger) per 100 ml | None identified | Specialized synthesis and handling equipment to maintain purity of the materials | None identified | WA IL Cat 1 MTCR 9, 11 |